# Robust Control Design of an Induction Motor

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Abstract—A robust controller for an AC induction motor is designed using principal gains method to govern stator currents. State space equations of an AC induction motor in  $\alpha$ - $\beta$  stator fixed frame are nonlinear with respect to rotor speed. Nonlinear equations are rewritten to parameter from describing dependency on rotor speed which is assumed to be known by measurement or by estimation. A nominal model ( $\omega$ =0rd/s) is considered for synthesis, and all regimes dependent of  $\omega$  are considered perturbed regimes at output multiplicative uncertainty. The singular values of uncertainty are quantified. Robust conditions of stability and performances are given. It is showed, from the results in frequency and time domain, that the principal gains method can be successfully applied to induction motor.

*Index Terms*—principal gains, singular values, condition number, AC induction motor

#### I. INTRODUCTION

The standard modern approach to induction motor control consists in constructing a mathematical model and than using explicitly this model in the controller. However, there are two major problems with this approach: first, the model is only a simplified representation of the dynamic AC induction motor which is generally more complex; second, the AC induction motor behavior continuously changes. For these two reasons there is inevitably a mismatch between the plant and the model. Such model uncertainties are responsible for the degradation of the controller. Hence, the first step in robust control study is to quantify these uncertainties. For that purpose, a simplified model of an AC induction motor ( $\omega$ =0rd/s) and different models depending of the frequency are used to design a robust controller with principal gains method.

## II. PRELIMINARIES

It is necessary to recall the basic required performances of a control loop in frequency domain. Fig. 1 shows the classical structure of a control loop with the main components: the controller (transfer matrix K(s)), the process (transfer matrix G(s)), the multiplicative

model uncertainty at the process output  $\Delta_m(s)$ , the setpoint r, the loop's error e and finally the manipulated variable u and output y. let G'(s) the transfer matrix of the true plant, all perturbed regimes, then the following relation can be written:

$$\mathbf{G}'(\mathbf{s}) = [\mathbf{I} + \Delta_m(\mathbf{s})] \mathbf{G}(\mathbf{s}) \tag{1}$$

The largest singular value of  $\Delta_m(s)$  is obtained from "(1)":

$$\sigma_{max} \left[ \Delta_m(s) \right] = \sigma_{max} \left( \left[ G'(s) - G(s) \right] G^{-1}(s) \right)$$
 (2)

"Equation (2) is used to quantify the multiplicative models uncertainties".



Figure 1. Feedback configuration with multiplicative uncertainties

#### A. Robust Stability

Assume that the nominal feedback system G(s) (i.e. with  $\Delta_m(s) = 0$ ) is stable, then the true feedback system G'(s) is stable if the following inequality holds [1]:

$$\sigma_{max} \left[ T(s) \right] < 1 / \sigma_{max} \left[ W_t(s) \right]$$
(3)

where T(s) is the nominal closed loop transfer matrix given by:

$$T(s) = G(s) K(s) [I + G(s) K(s)]^{-1}$$
(4)

And W<sub>t</sub>(s) is stability specification matrix such as:

$$\sigma_{max} \left[ \Delta_m(\mathbf{s}) \right] \le \sigma_{max} \left[ \mathbf{W}_{\mathsf{t}}(\mathbf{s}) \right] \tag{5}$$

Then the maximum principal gains  $\sigma_{max}$  [T(s)], the largest singular value of the nominal closed loop transfer matrix is a reliable indicator of the robust stability of the feedback system. "Equation (3) is the robustness condition of the feedback system".

## B. Robust Performances

Let  $W_p$  (s) a performance specification matrix, weighting matrix, than the robust performances of all

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perturbed regimes G'(s) are satisfied if the following inequality holds [1], [2], [3]:

$$\sigma_{max} [S(s)] \le 1/\sigma_{max} [W_p(s)] \tag{6}$$

where S(s) is the sensitivity matrix given by:

$$S(s) = [I + G(s)K(s)]^{-1}$$
 (7)

In fact, the largest singular value of the sensitivity matrix  $\sigma_{max}$  (s) is also an indicator of the sensitivity of the system response to a change of the plant character.

In conclusion, the inequalities "(3)" and "(6)" represent the robustness conditions and must be satisfied to obtain a robust controller.

#### III. PRINCIPAL GAINS METHOD

The principal gains method is based on finding a controller with the following structure [3]:

$$K(s) = K1*K2(s)*K3*K4(s)$$
 (8)

where:

 $K1=G^{-1}$  (0) is the inverse static gain. It is used to decouple the process in low frequency.

K2(s) = I/s is a set of integrators to eliminate the static error.

K3 is a compromise coefficient between the stability and performances.

K4(s) is a structure to reduce the resonance magnitude in middle and high frequency. In order to not affect the controller in low frequency, we have to set K4 (0) =I, this can be obtained by minimization of the following criteria [3], [4]:

$$\min_{K_4}(J) = \min_{K_4}\max_{W} \left[\sigma_{max}(T) \sigma_{max}(\Delta_m)\right] \quad (9)$$

where:  $\sigma_{max}(T)$ ,  $\sigma_{max}(\Delta_m)$  is a stability robust condition.

## IV. APPLICATION TO THE AC INDUCTION MOTOR

The state space equations of an AC induction motor dependent on  $\omega$ = [-110rd/s, 110rd/s] are given in [5] and [6]:

$$\dot{\mathbf{x}} = (\mathbf{A}_0 + \boldsymbol{\omega} \mathbf{A}_1) \mathbf{x} + \mathbf{B}\mathbf{u}$$
$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \tag{10}$$

With state vector  $\mathbf{x} = [\varphi_a, \varphi_b, i_a, i_b]^T = [\mathbf{x}_1, \mathbf{x}_2, \mathbf{x}_3, \mathbf{x}_4]^T$ , where  $\varphi_a, \varphi_b$  are the rotor fluxes and  $i_a, i_b$  are the stator currents and the control input  $\mathbf{u} = [\mathbf{u}_1, \mathbf{u}_2]^T$  represents the stator voltages. The measured output is  $\mathbf{y} = [\mathbf{i}_a, \mathbf{i}_b]^T$ .

The matrices  $A_0$ ,  $A_1$ , B, C and D are given as:

$$A_{0} = \begin{pmatrix} a_{1} & 0 & a_{2} & 0\\ 0 & a_{1} & 0 & a_{2}\\ a_{3} & 0 & -\gamma & 0\\ 0 & a_{3} & 0 & -\gamma \end{pmatrix}$$
(11)

$$A_{1} = \begin{pmatrix} 0 & -n_p & 0 & 0 \\ n_p & 0 & 0 & 0 \\ 0 & a_4 & 0 & 0 \\ -a_4 & 0 & 0 & 0 \end{pmatrix}$$
(12)

$$\mathbf{B} = \begin{pmatrix} 0 & 0 \\ 0 & 0 \\ a_5 & 0 \\ 0 & a_5 \end{pmatrix}, \quad \mathbf{C} = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix} \quad (13)$$

where the following parameters are given in Table I:

$$a_1 = -\frac{1}{T_r}, a_2 = \frac{L_{sr}}{T_r}, a_3 = \frac{L_{sr}}{T_r \sigma L_s L_r}, a_4 = \frac{n_p L_{sr}}{\sigma L_s L_r}, a_5 = \frac{1}{\sigma L_s},$$
$$T_r = \frac{L_r}{R_r}, \gamma = \frac{R_s}{L_s \sigma} + \frac{L_{sr}^2}{L_s \sigma L_r T_r}, \sigma = 1 - \frac{L_{sr}^2}{L_s L_r}$$

 
 TABLE I.
 The Nominal Physical Parameters of the Induction Motor

Description	Parameter	Value	Units
Stator Inductance	Ls	0.47	Н
Rotor Inductance	Lr	0.47	Н
Mutual Inductance	L <sub>sr</sub>	0.44	Н
Leakage factor $\sigma_s = \sigma_r$	σ	0.12	
Stator resistance	R <sub>s</sub>	0.8	Ω
Rotor resistance	R <sub>r</sub>	3.6	Ω
Moment of inertia	$D_m$	0.06	$Kg.m^2$
Viscous damping constant	R <sub>m</sub>	0.04	N.m.s
Number of pole pairs			
_	$n_p$	2	

The step responses for -110 rd/s  $\leq \omega \leq 110$  rd/s are represented in Fig. 2 and Fig. 3:



Figure 2. Open loop time responses for  $\omega$ = [-110rd/s, 110rd/s]. Unit step demand in u<sub>1</sub>



Figure 3. Open loop time responses for  $\omega$ = [-110rd/s, 110rd/s]. Unit step demand in u<sub>2</sub>

From the Fig. 2 and Fig. 3 it is clearly that there are cross-couplings and badly damped modes whose frequencies vary considerably with the rotor speed. This coupling is also showed in Fig. 4 because the condition number [7]-[9], is between 2 and 8.



## A. Evaluation of Multiplicative Uncertainties $\Delta_m(s)$

The largest singular values of the multiplicative uncertainties  $\Delta_m(s)$  are determined from "(2)".

The result is given in Fig. 5, where it is verified that the norms of these uncertainties are less than one at low frequencies and increase at high frequencies [1].

#### B. Robustness Conditions

Using "(5)" and the result given in Fig. 5, the stability specification matrix  $W_t(s)$  is represented as:

$$W_t(s) = \begin{pmatrix} 0.8(1+1.5s) & 0\\ 0 & 0.8(1+1.5s) \end{pmatrix}$$
(14)

Then, the condition for robust stability is given by "(3)". The performance specification for all possible plants, perturbed regimes, are defined such these regimes have the same response time that the nominal regime when the pulsation  $\omega = 0$  rd/s, then the nominal dynamic matrix is A<sub>0</sub>. Then, the performance specification matrix  $W_p(s)$  is given by:

$$W_p(s) = \begin{pmatrix} \frac{2s+1}{2s} & 0\\ 0 & \frac{2s+1}{2s} \end{pmatrix}$$
(15)

The condition for robust performance is given by "(6)". Finally, the robustness conditions for AC induction motor are represented in Fig. 6.

## C. Robust Controller with Principal Gain Method for AC Induction Motor

The controller is given by "(8)" where:

K1 = G<sup>-1</sup>(0) = 
$$\begin{pmatrix} 0.8 & 0 \\ 0 & 0.8 \end{pmatrix}$$
; K2(s) =  $\frac{l}{s} = \begin{pmatrix} \frac{1}{s} & 0 \\ 0 & \frac{1}{s} \end{pmatrix}$ 

K3 = 0.6 is determined by simulation.

K4(s) is obtained by minimization of the criteria "(9)":

$$K4(s) = \begin{pmatrix} 1 + 0.07261s & 0\\ 0 & 1 + 0.07261s \end{pmatrix}$$

Finally the controller is:

$$K(s) = \begin{pmatrix} \frac{0.04066s + 0.56}{s} & 0\\ 0 & \frac{0.04066s + 0.56}{s} \end{pmatrix}$$
(16)

The results in frequency domain are given in Fig. 7, where it is showed that the robustness condition are not violated because, for multivariable system, the stability is guaranteed if the largest singular values of the closed loop transfer matrix ( $\sigma_{max}[T(s)]$  is lower than the upper bound of the largest singular values of the model uncertainties ( $1/\sigma_{max}[W_t(s)]$ ). The same idea is used for the robust performance criterion.



Figure 5. Multiplicative uncertainties  $\Delta m(s)$ 



Figure 6. Robustness conditions of stability and performances.



Figure 7. Results in frequency domain





In Fig. 8 and Fig. 9 the results in temporal domain are given; the stability of all regimes and good performances which means small interactions, are observed.



Figure 9. Step responses: echelon u2.

## V. CONCLUSION

This article presented principal gains method to design robust controller for an AC induction motor. It is showed, from the results in frequency and time domain, that the principal gains method can be successfully applied to induction motor. The dynamic behavior of induction motor is generally difficult to model, it could be done. The theory behind the robust control tools is simplified to be easily transmitted to electrical processing students and engineers.

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